

Foldcore Sandwich Structures for Hypervelocity Impact Energy Absorption Applications under Cold Temperatures

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USU MAE Manufacturing/Materials Research



Dr. Thomas Fronk
Composite Structures
Cryogenic Applications
Numerical Modeling



Dr. Juhyeong Lee
Composite Structures
Fiber/Composite Manufacturing
Computational Mechanics
Finite Element Modeling

Juhyeong Lee

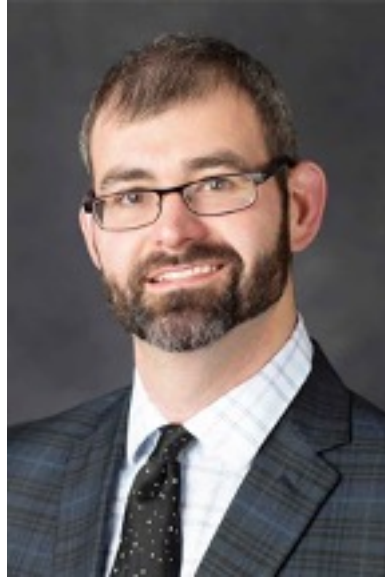


Dr. Nadia Kouraytem
Laser-Based Metal AM
Material Characterization
Nuclear Materials

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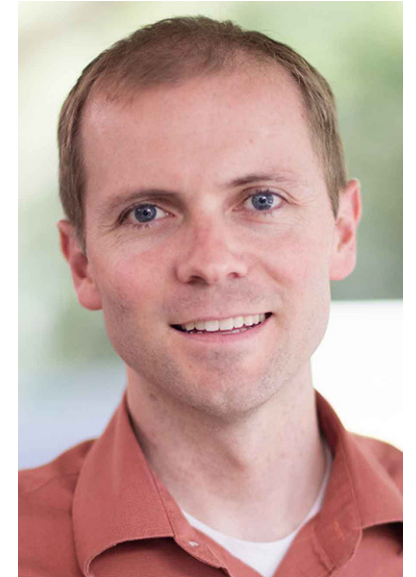
Dr. Nick Roberts
Thermal Energy
Phase Change Materials
Nano-fluids/fabrication



Dr. Ryan Berke
Solid Mechanics
Advanced Materials
Experimental Methods



Dr. Tony Whitmore
Propulsion
Arc-Ignition System
Micro-Hybrid Gas Generator

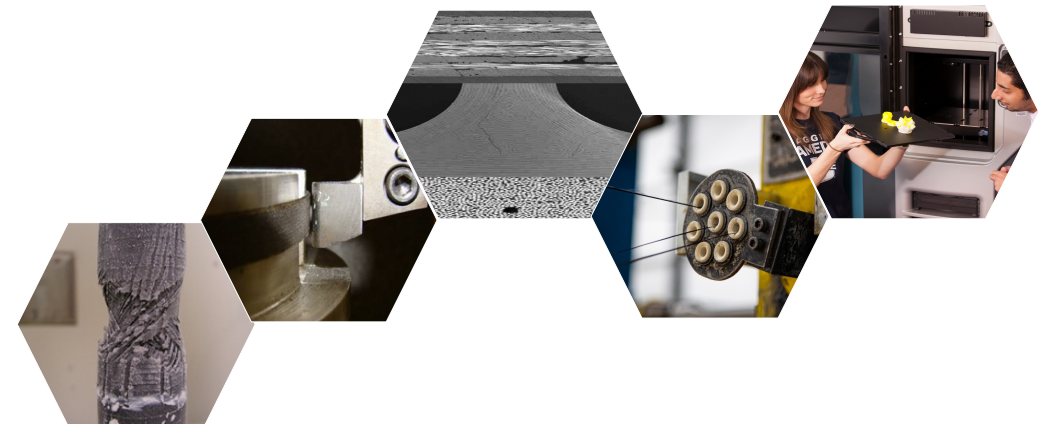
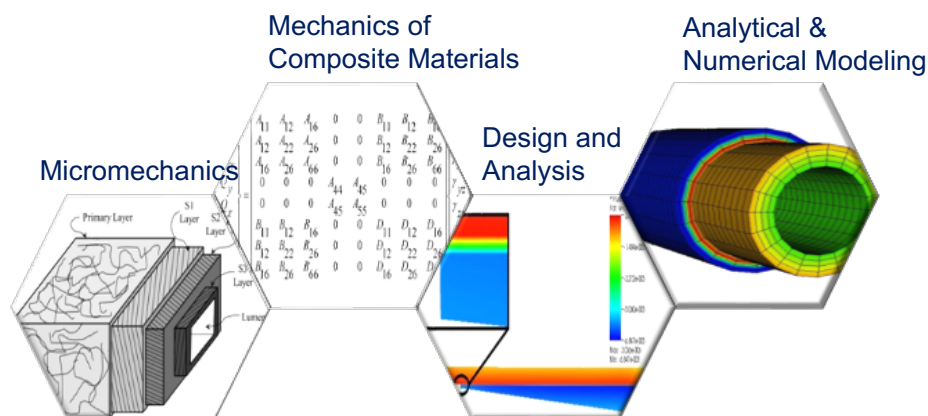


Dr. Doug Hunsaker
Aerodynamic Design
Morphing Aircraft
Subsonic/Supersonic Flight

USU Composites Research

Center for Manufacturing & Design of Advanced Materials

- MS in Composite Materials & Structures (USU/WSU joint graduate program)
- **Objective:** To provide a structure for education and research to improve the design and processing of advanced materials in the industries in the state of Utah
- **Outcomes:** Increased number of prepared students to enter industries in the state. Improved manufacturing techniques through research.



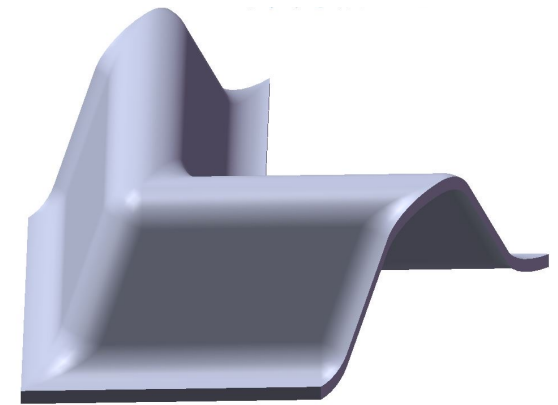
Foldcore – Background

A foldcore is an alternative core material made of thin carbon or glass fiber laminates folded in an origami pattern, providing tunable structural performance.

Foldcore Sandwich Composites (FSCs) have the following potential advantages over legacy core materials:

- Open channels for moisture evacuation and heat transfer
- Optimizable unit cell geometry
- Redirection of impact debris
- Core material is stronger than paper honeycomb or foam

The tunable nature of a foldcore design provides versatility and application-specific properties.



Foldcore unit cell
(Miura-folding)

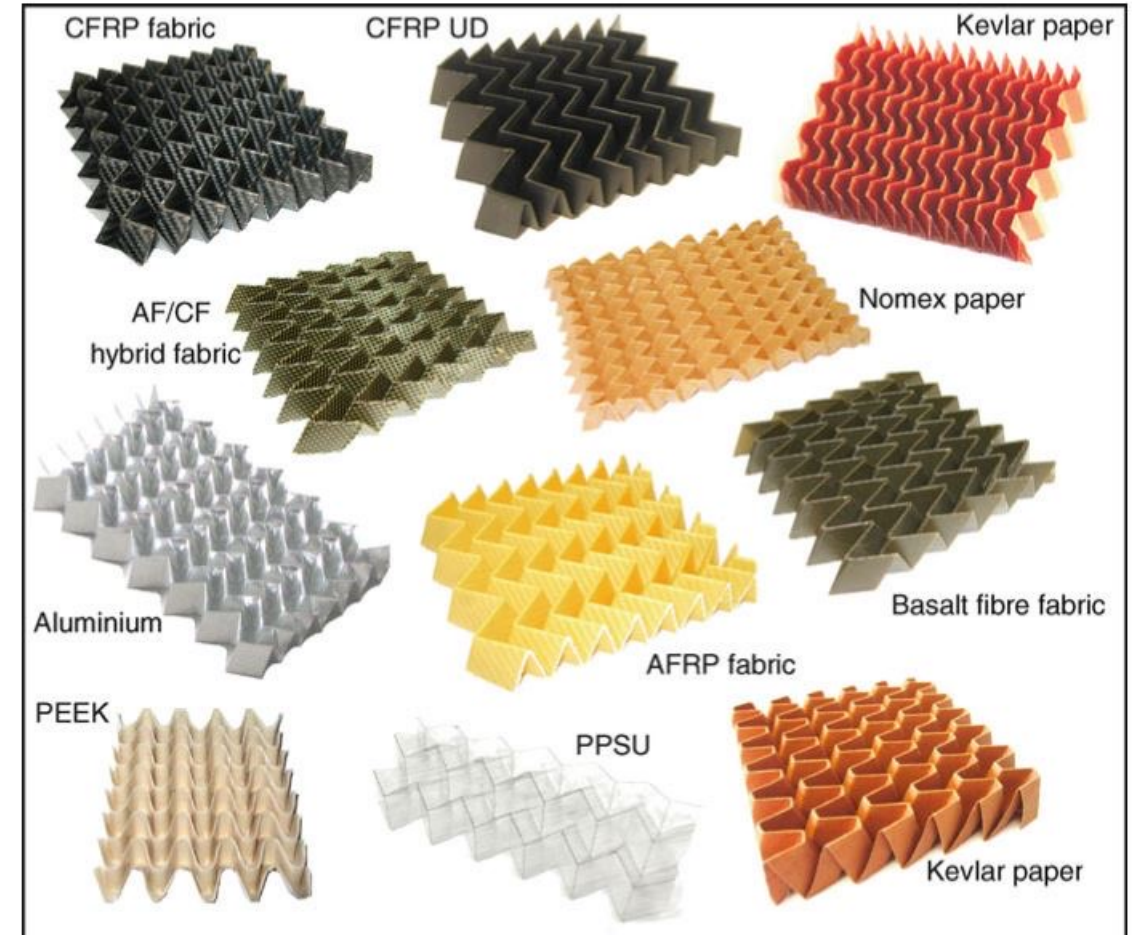
Foldcore – Background

Substantial work has been done on foldcore research since the 1990's.

Foldcores have been manufactured with various materials and shapes. Papers have been published on the following mechanical tests.

- Compression/Shear
- Bending & Skin Debonding
- Low-High Velocity Impacts

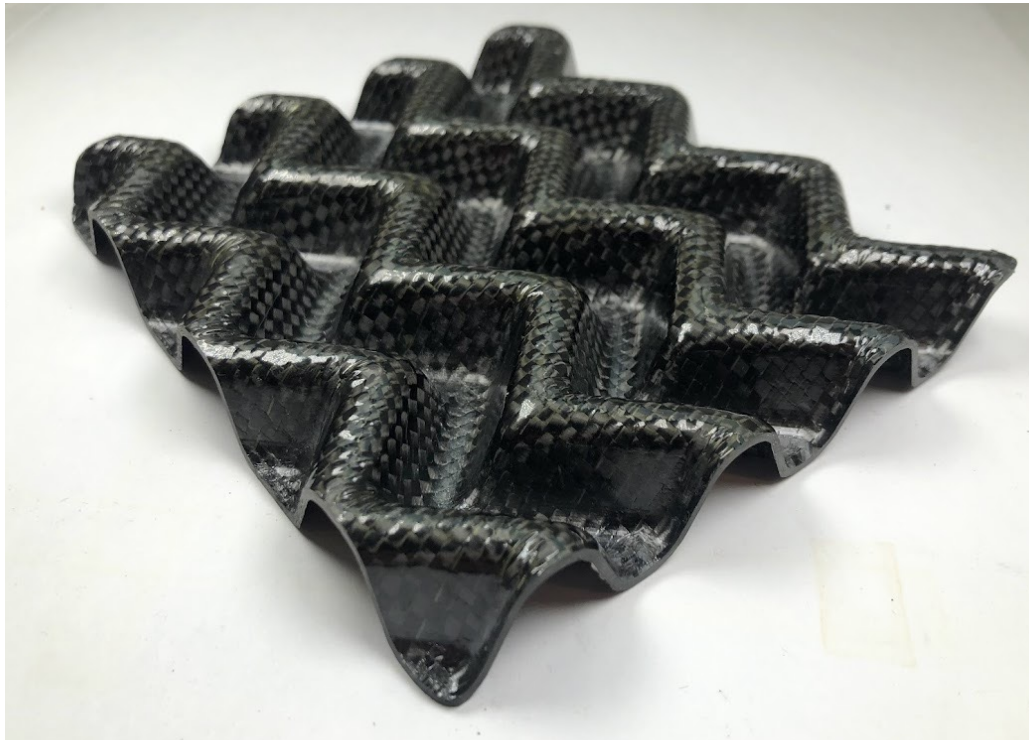
Potential applications include, aircraft fuselages, control surfaces, marine structures as protective armor for military and space systems.



Heimbs (2012), DFCSS, 491–544.

Foldcore Manufacturing

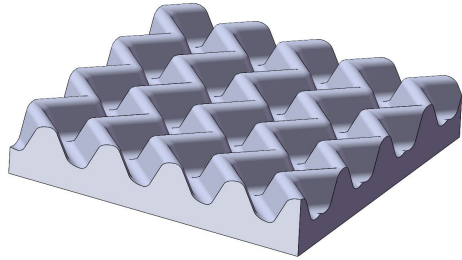
USU started preliminary study in Fall 2021 and secures FSC manufacturing capability early 2022



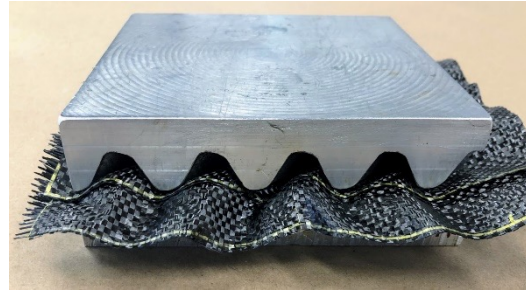
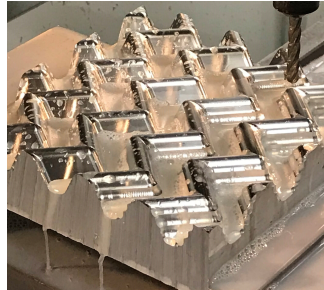
CFRP Foldcore

- Miura-folding
- 5" x 5" x 0.7" ($t = 0.06$ ")
- Miura-folding pattern
- 8 woven carbon/epoxy prepregs.
 - Hexcel AS4 3k PW CF fabric
 - 3501-6 epoxy resin

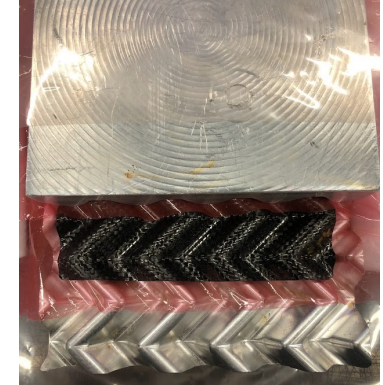
WF prepregs shows better formability than UD one.



Mold fabrication



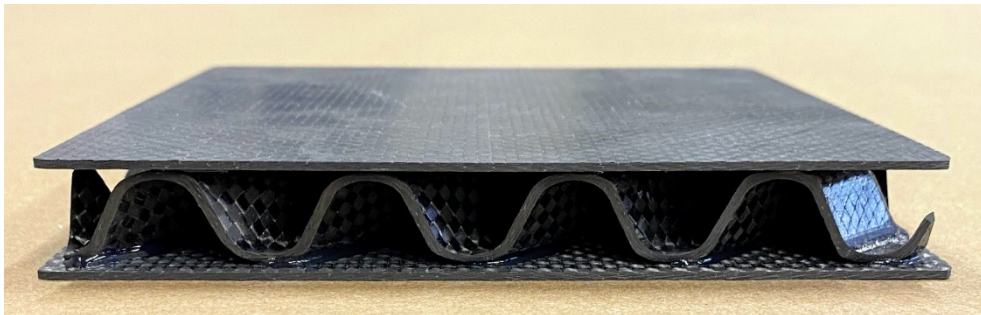
Pre-folding



Lay-up & bagging



Cure



CFRP FSC

Foldcore and skin panels are bonded with 3M DP420 epoxy adhesive.

(90°)

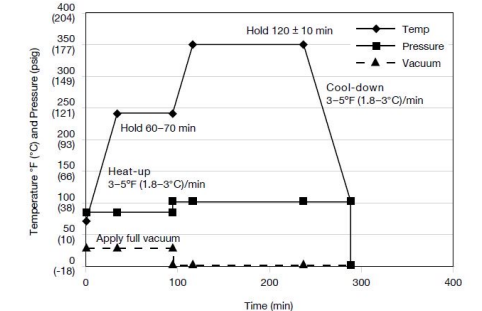


CFRP Foldcore

Open Channel (0°)



Cure Cycle

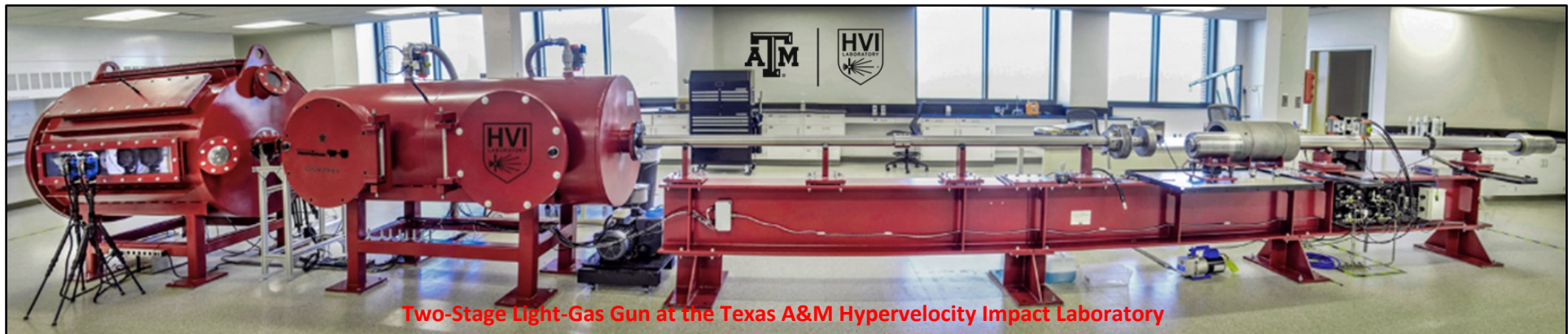


Hypervelocity Impact Tests

USU collaborates with TAMU Hypervelocity Impact Lab for “development and tailoring of novel structures comprised of advanced materials, including metals, composites, polymers, ceramics, and geomaterials.”

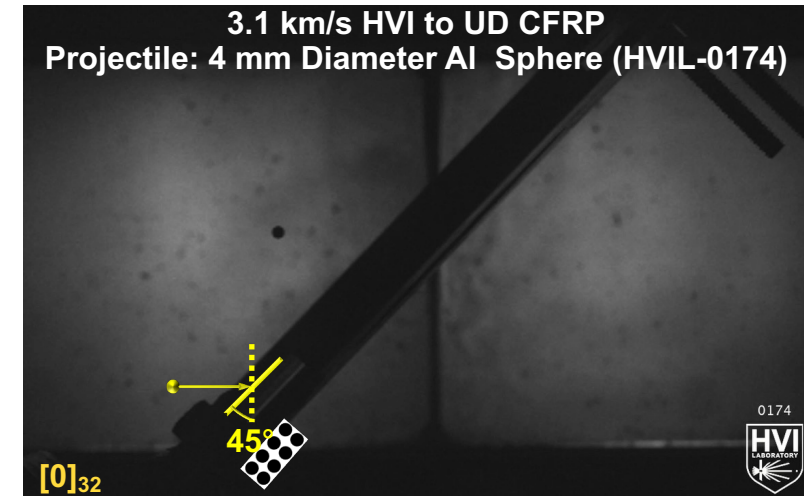
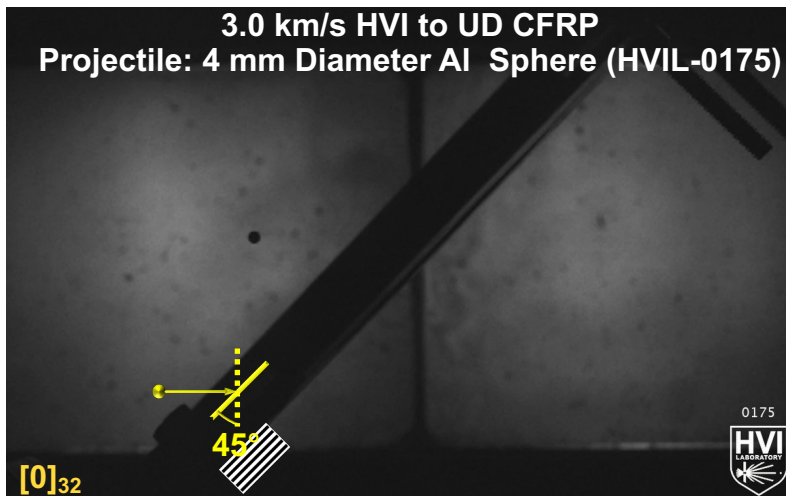
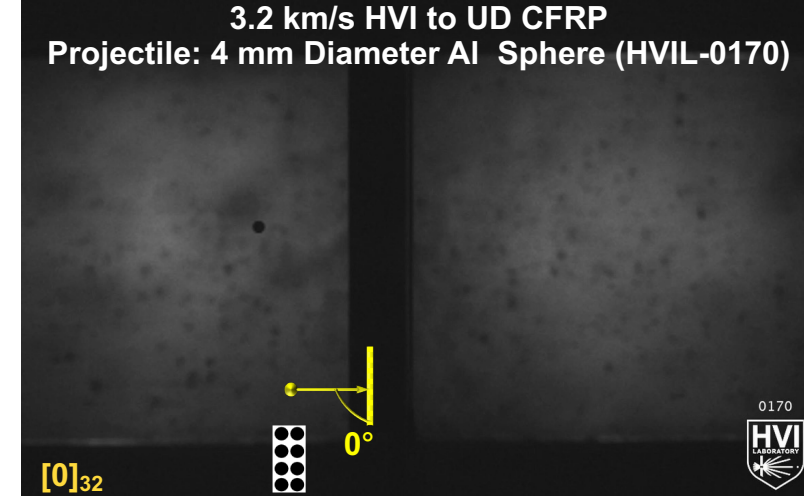
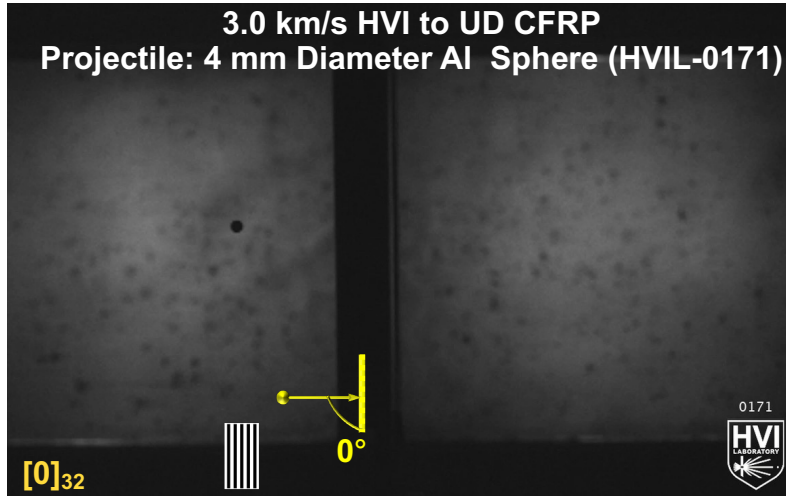
A series of HVI tests are performed:

- on flat CFRP panels with various layups (**CHECK** anisotropic failure).
- on CFRP FSCs with different opening channels (**CHECK** fragments redirections).
- at 3 km/s normal (0°) and oblique (45°) impact (**CHECK** fragments expansion).



Two-Stage Light-Gas Gun at the Texas A&M Hypervelocity Impact Laboratory

HVI Testing on Flat CFRP Panels

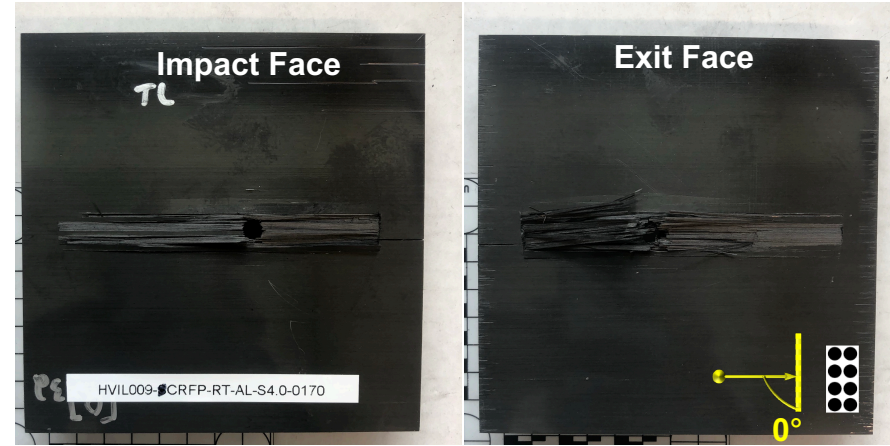


HVI Testing on Flat CFRP Panels

Normal 3.0 km/s impact, $[0]_{32}$



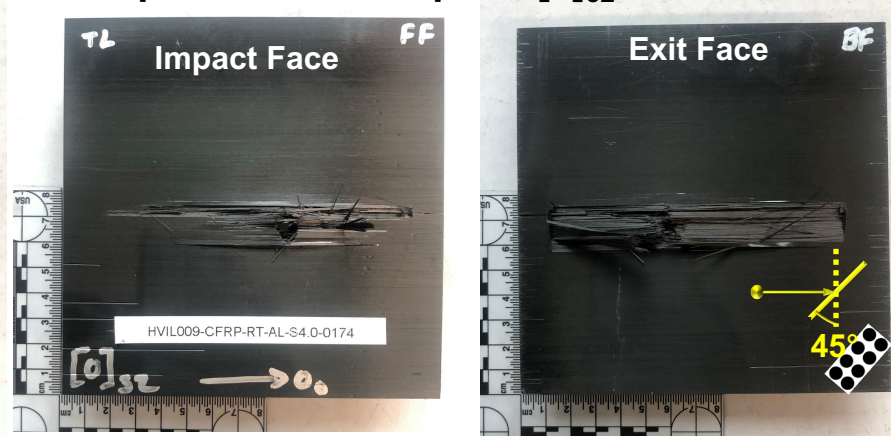
Normal 3.2 km/s impact, $[0]_{32}$



Oblique 3.0 km/s impact, $[0]_{32}$



Oblique 3.1 km/s impact, $[0]_{32}$



Key Findings:

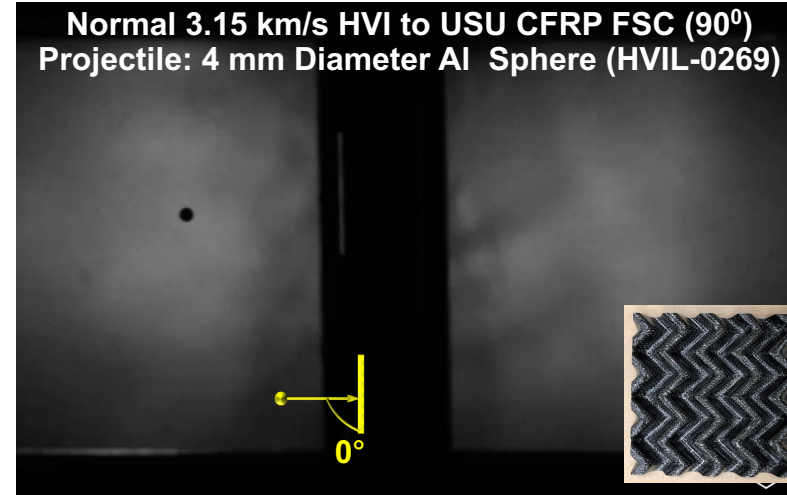
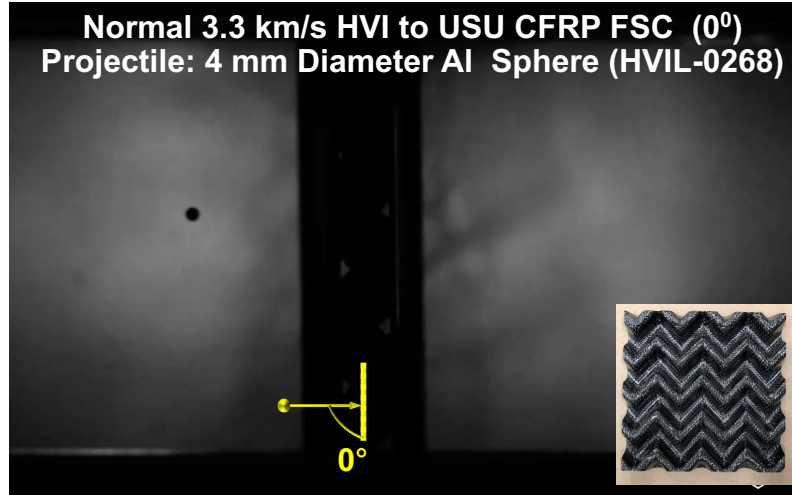
Fragment cloud expansion and structure damage are governed by:

- Lay-up
- Impact angle

This suggests that **composite lay-up and impact angle** are design factors that can optimize HVI mitigation.

These can be easily tuned in FSCs!!

HVI Testing on CFRP FSCs

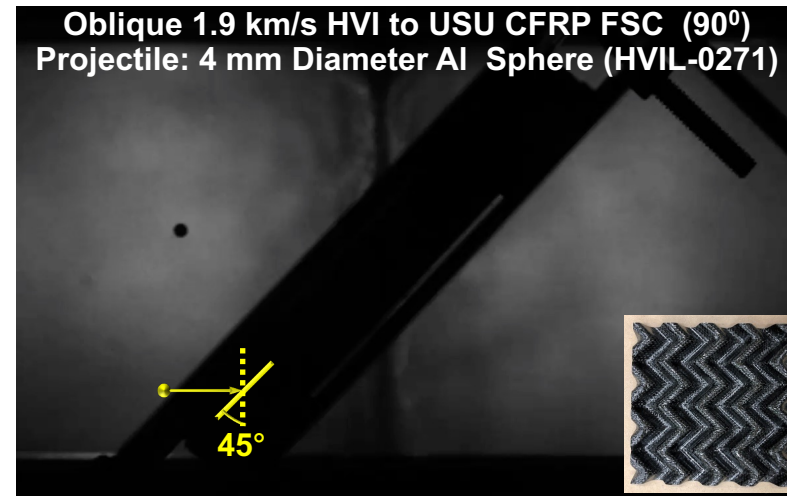
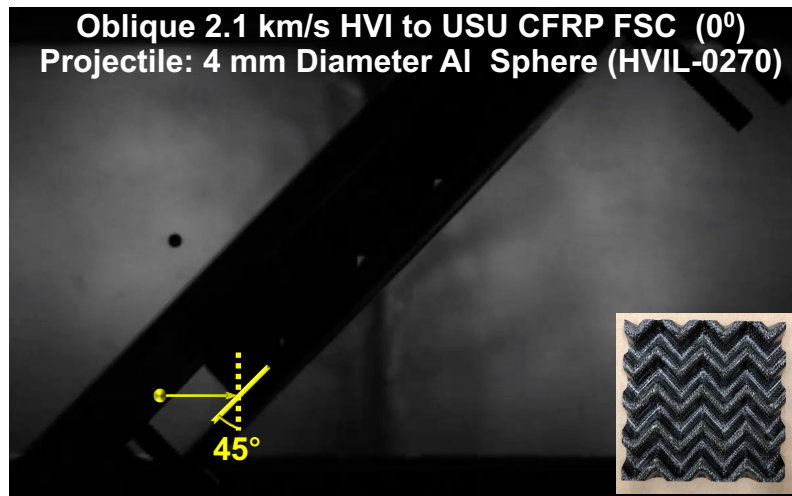


Validation:

Fragment cloud expansion and structure damage are very different for flat and FSC CFRPs

Foldcore configurations (i.e., material, layup, folding pattern, type) can be optimized for the maximum HIV resistance.

Typical honeycomb & polymer sandwich panels have *limited* design variables.

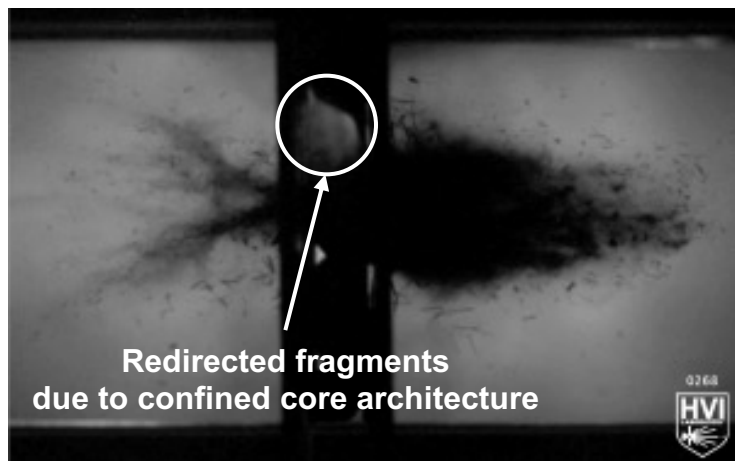


Why FSC Performs Better?

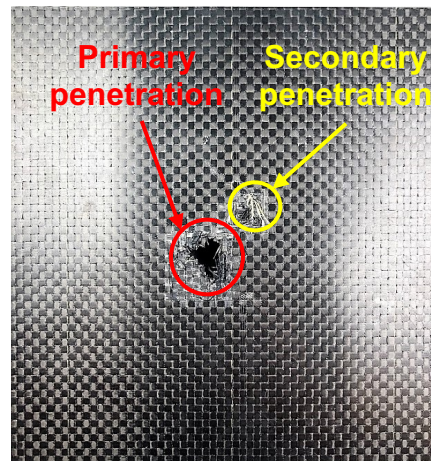
FSC has several open channels. Each of open channel creates internal bumper and redirects fragments inside the foldcore, creating less back-face fragment cloud expansion and smaller exit hole, i.e., the Whipple shield for spacecraft protection.

Prototype FSC under normal impact (left) shows two exit holes, suggesting the potential controllability of a projectile. *Ideally*, optimal FSC can capture/hold an incoming projectile.

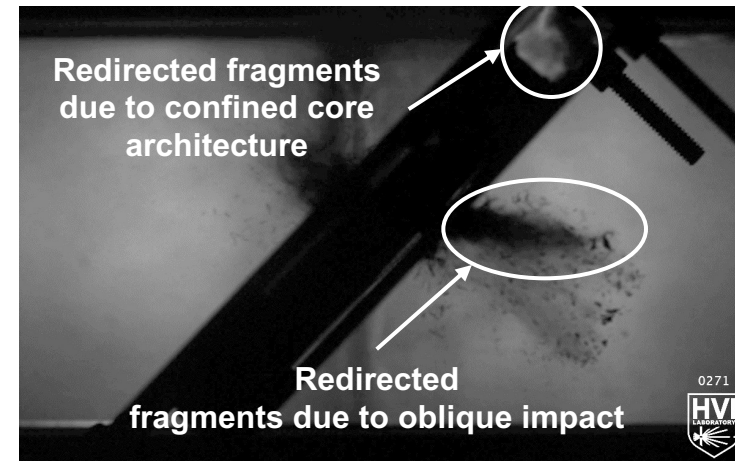
Fragments Cloud Expansion



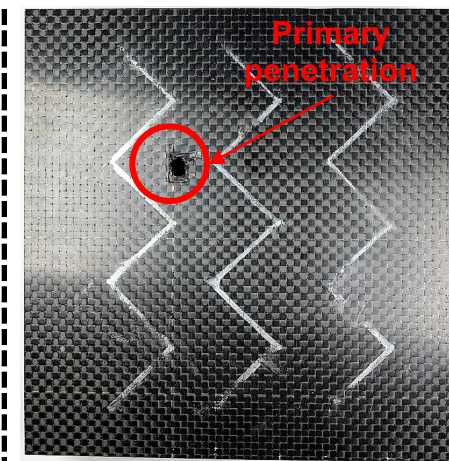
Back-face Damage



Fragments Cloud Expansion



Back-face Damage



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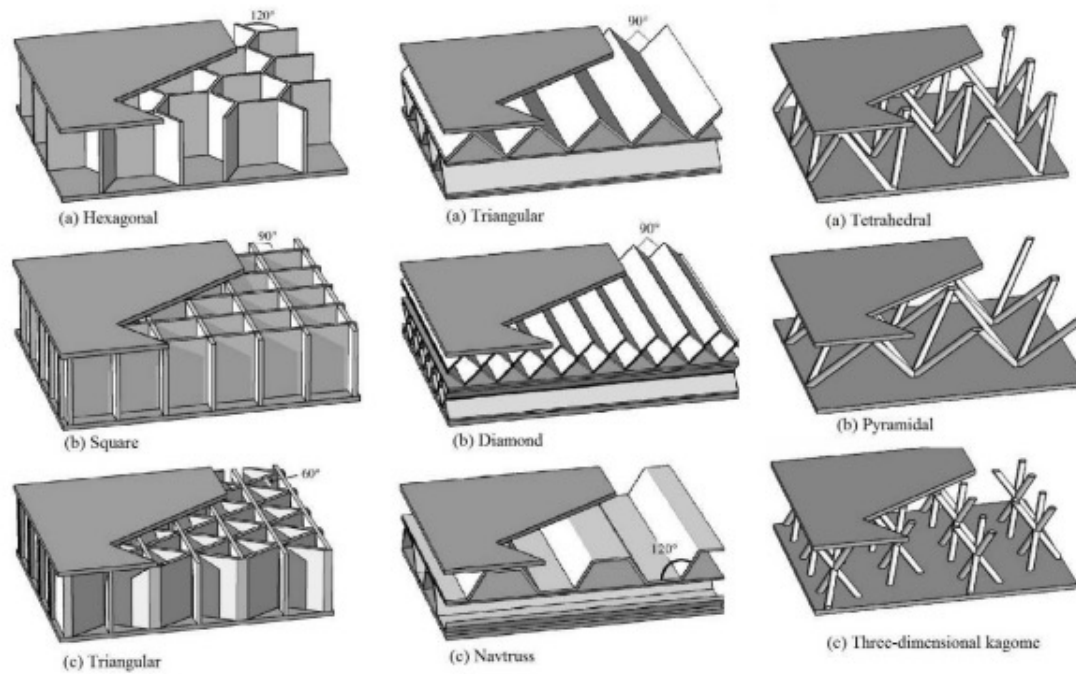
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FSC Design Variables – Core Topologies

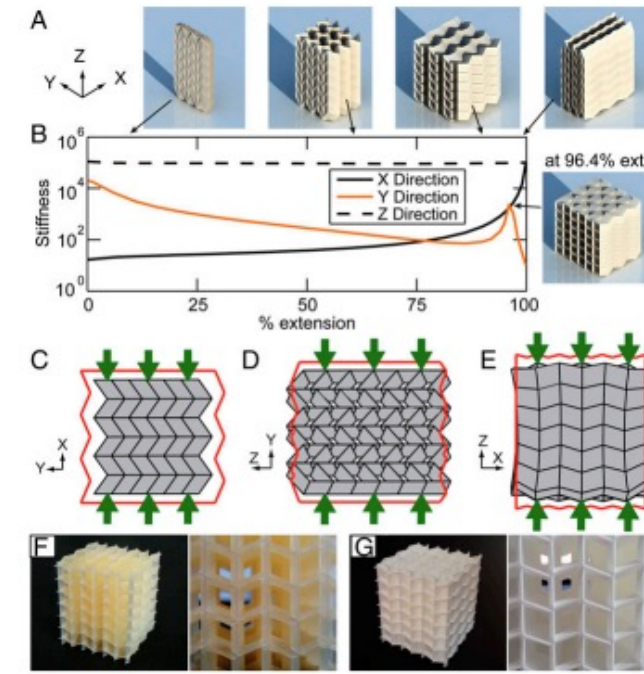
FSC performance can be easily tailored to given design requirements/constraints.

Geometric Variables:

- Folding pattern, unit-cell geometry, stacking pattern/sequence, etc.



Wadley (2006), Philos. Trans. R. Soc. A, 31-68.

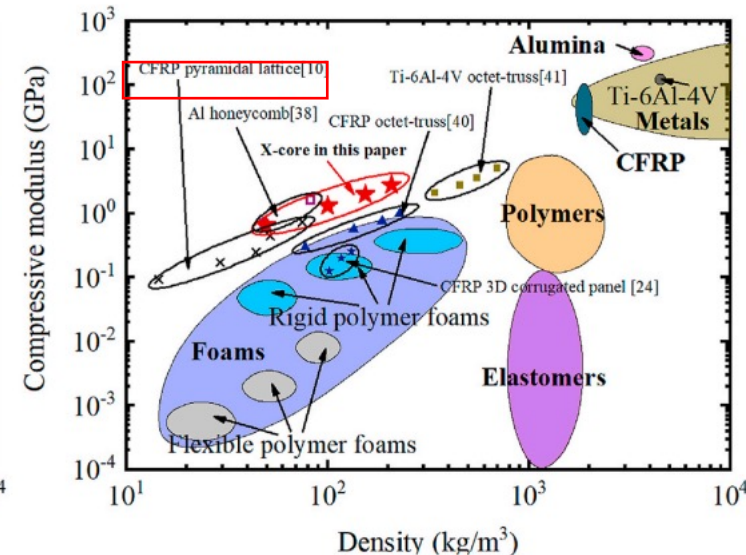
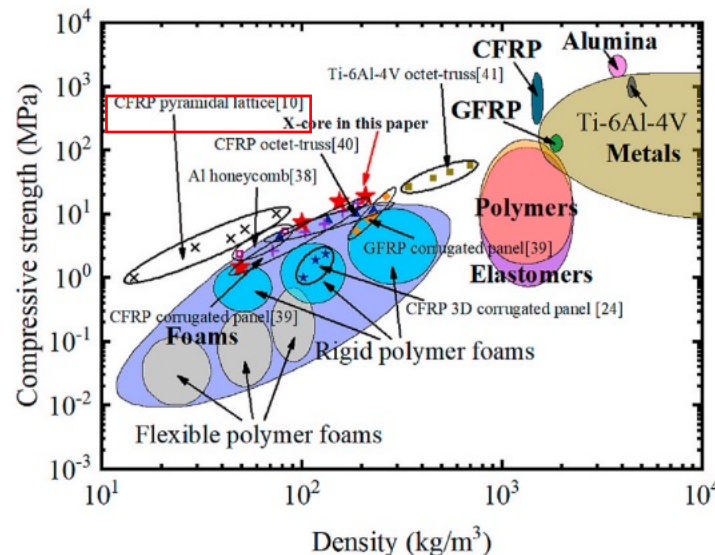


Filipov, Tachi, & Paulino (2015), PNAS, 12321–12326.

FSC Design Variables – Materials

Material Variables:

- A sheet of single or multi-layered materials can be folded in any shapes.
 - Aluminum, CFRP, GFRP, Aramid, Kevlar, Basalt, PEEK, etc.
 - Multi-layered hybrid foldcores often manufactured for multifunctionalities.
 - Manufacturability: prepregs >> dry fabrics
- Use of stiffer materials requires aggressive manufacturing process.



Mei et al (2021), TWST, 107144

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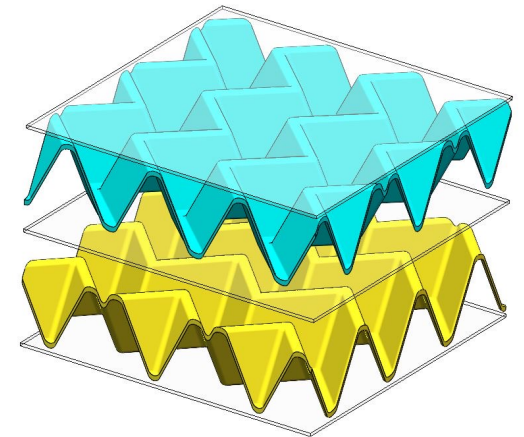
FSC at Extreme Environments

Significance of the Problem:

- Conventional foam cores have issues on limited geometric design variables to improve energy absorption, expensive manufacturing, and moisture/fluid accumulation.
- Sandwich structures may not perform well at extreme environments (i.e., temperature and water flow).
- Performance are hardly tailorable.

Opportunities:

- FSCs can be easily manufactured to have superior mechanical performance and several multifunctionalities (it is integrated structure) by:
 - Core topology optimization
 - Appropriate core material selection



USU FSC Research

Goals:

- The research is aimed at designing, testing, characterizing, and multiscale modeling novel lightweight FSCs, including both cores and face sheets to improve their HVI mitigation.
- The present work primarily focuses on characterizing the highly nonlinear dynamic HVI responses of FSCs with various origami-inspired core configurations (folding pattern, core dimensions, etc.) at low (-50°C), room, high ($50\sim 100^{\circ}\text{C}$) temperatures.

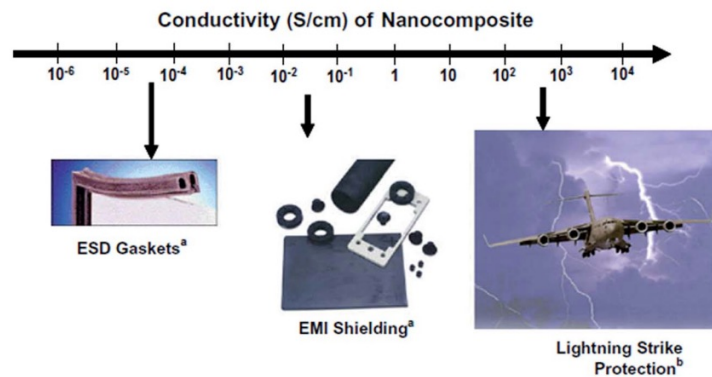
Objectives:

- To manufacture novel FSCs with tunable high energy impact resistance performance at various temperatures providing baseline data for future integrated research.
- To create a solid pipeline for USU students to work on “practical” research relevant to aerospace defense manufacturing.

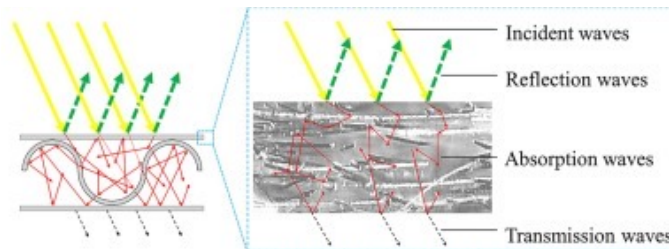
Current Status: RT HVI tests (done); low/high temp HVI tests (*proposed ideas*).

Potential Applications

Multifunctional structural panel with EMI/Lightning protection



www.chomerics.com & www.boeing.com

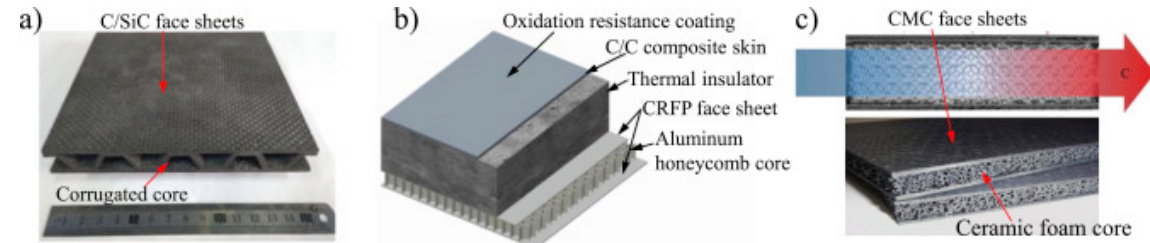


Liang et al (2021), JCOMA, 106481

Radar Absorbing Structure (RAS)

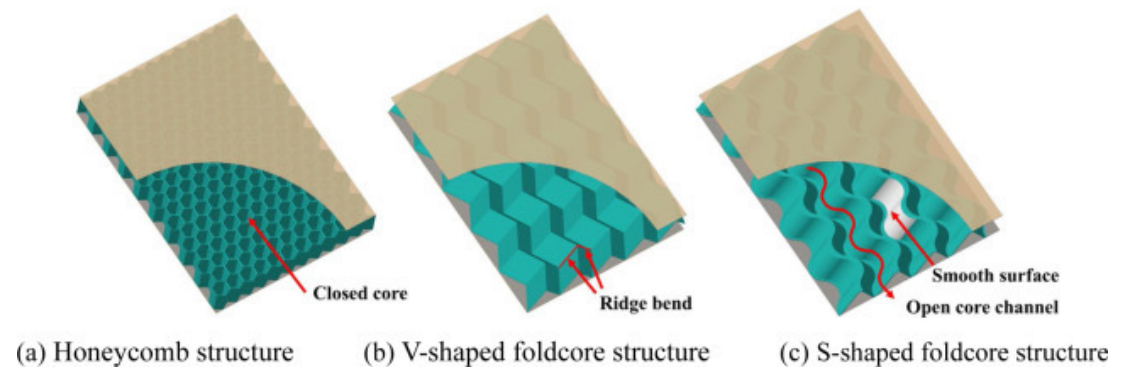
Thermal Protection System (TPS)

(foldcore is better than honeycomb, corrugated, lattice/truss core)



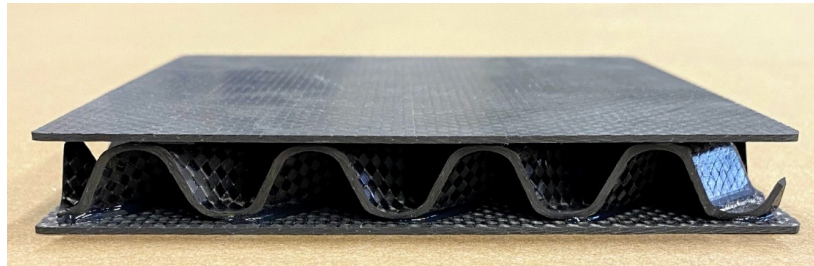
Le, Ha, Goo (2021), JCOMB, 109301

Thermal Management or Marine Structure



Deng et al. (2022), TWST, 109007

Questions?



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